

# On Interply friction in prepreg forming simulations



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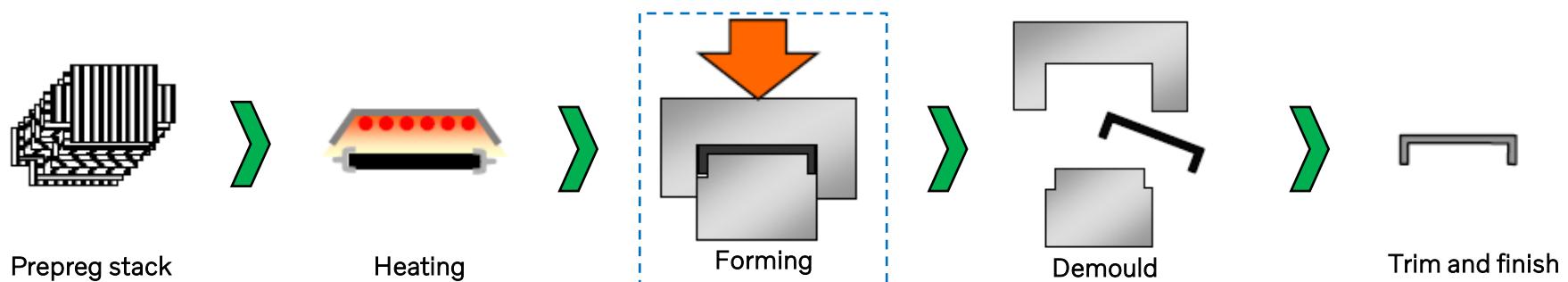
# Prepreg compression moulding

v o l v o

High level of automation

Higher volumes

Possible to co-process cheaper material systems. Eg, SMC

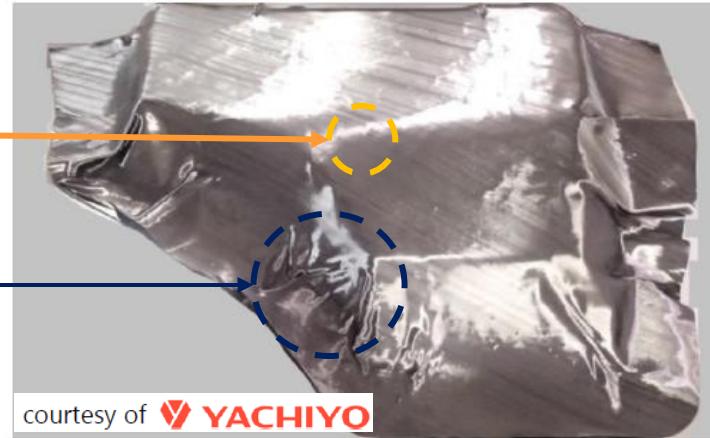


Courtesy of Mitsubishi Rayon

Defect

Fiber angle deviation

Wrinkles

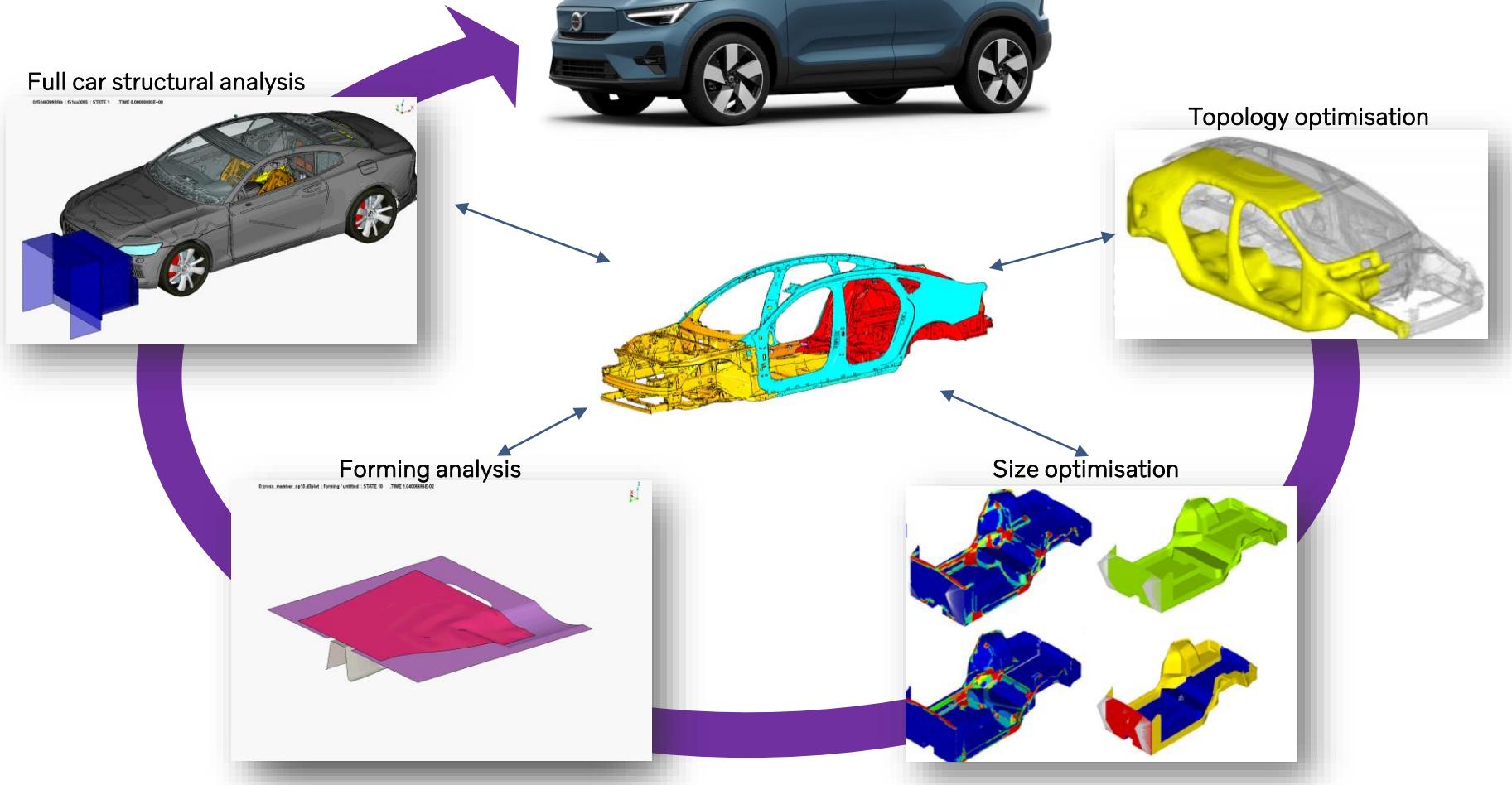


Process modelling is essential to predict and eliminate Forming induced defects

# Development cycle

V O L V O

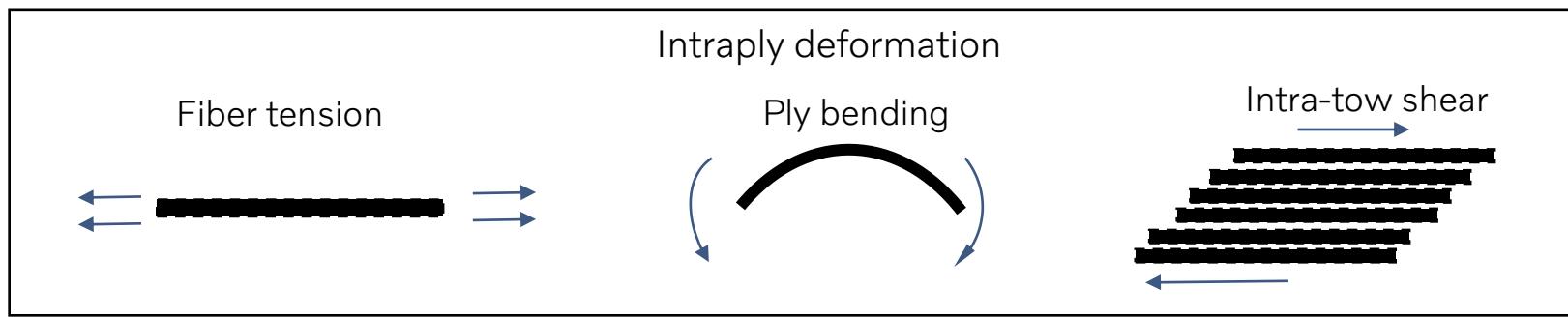
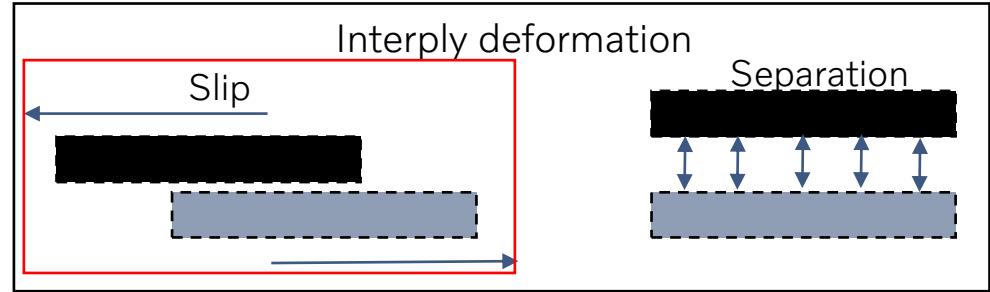
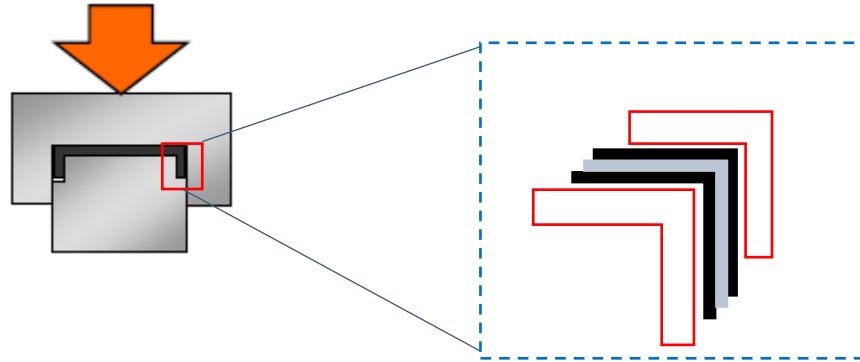
DYNA  
MORE  
NORDIC



# Deformation mechanism

v o l v o

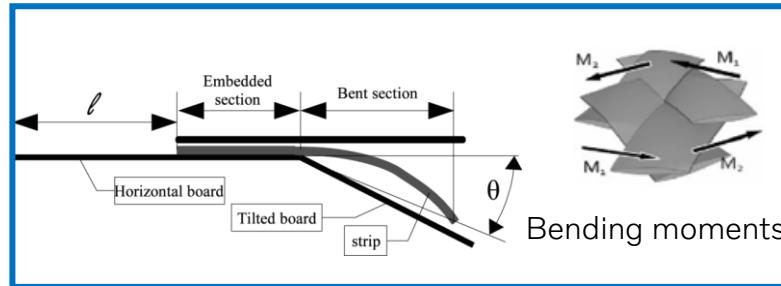
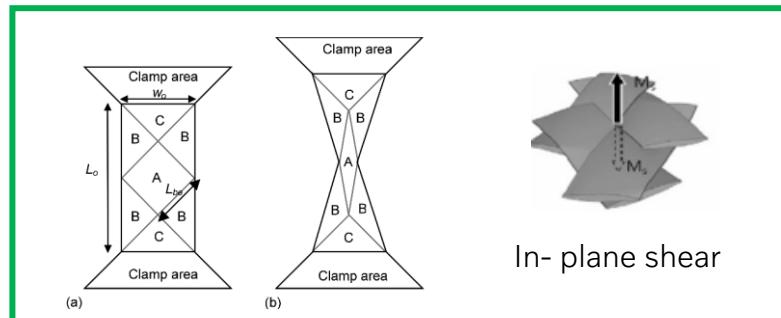
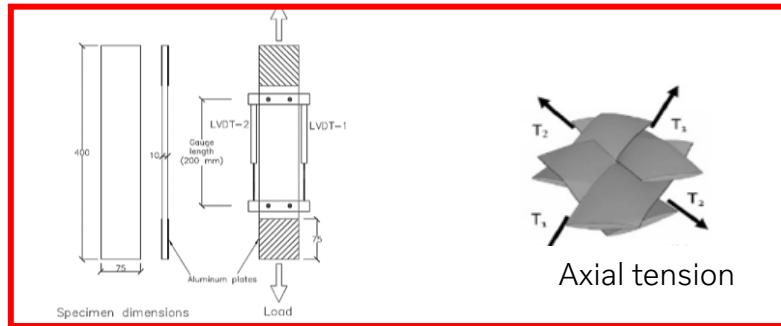
Continuous fiber reinforced prepgs



# Modelling Technique

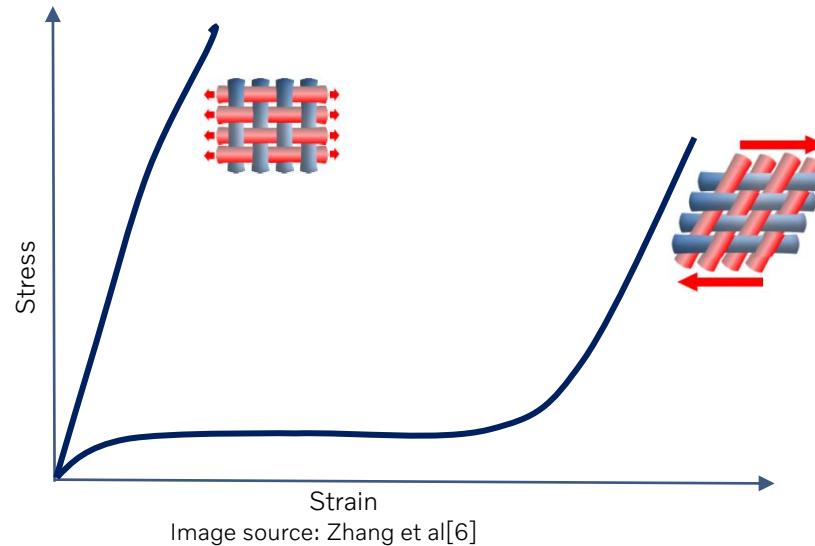


# Intraply deformation modes



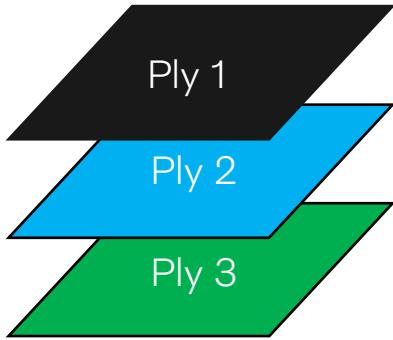
v o l v o

$$\begin{Bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \\ m_{11} \\ m_{22} \\ m_{12} \end{Bmatrix} = \begin{pmatrix} f1 & 0 & 0 & 0 & 0 & 0 \\ 0 & f2 & 0 & 0 & 0 & 0 \\ f12 & 0 & g1 & 0 & 0 & 0 \\ g1 & 0 & g2 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{pmatrix} \begin{Bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \gamma_{12} \\ \kappa_{11} \\ \kappa_{22} \\ \kappa_{12} \end{Bmatrix}$$



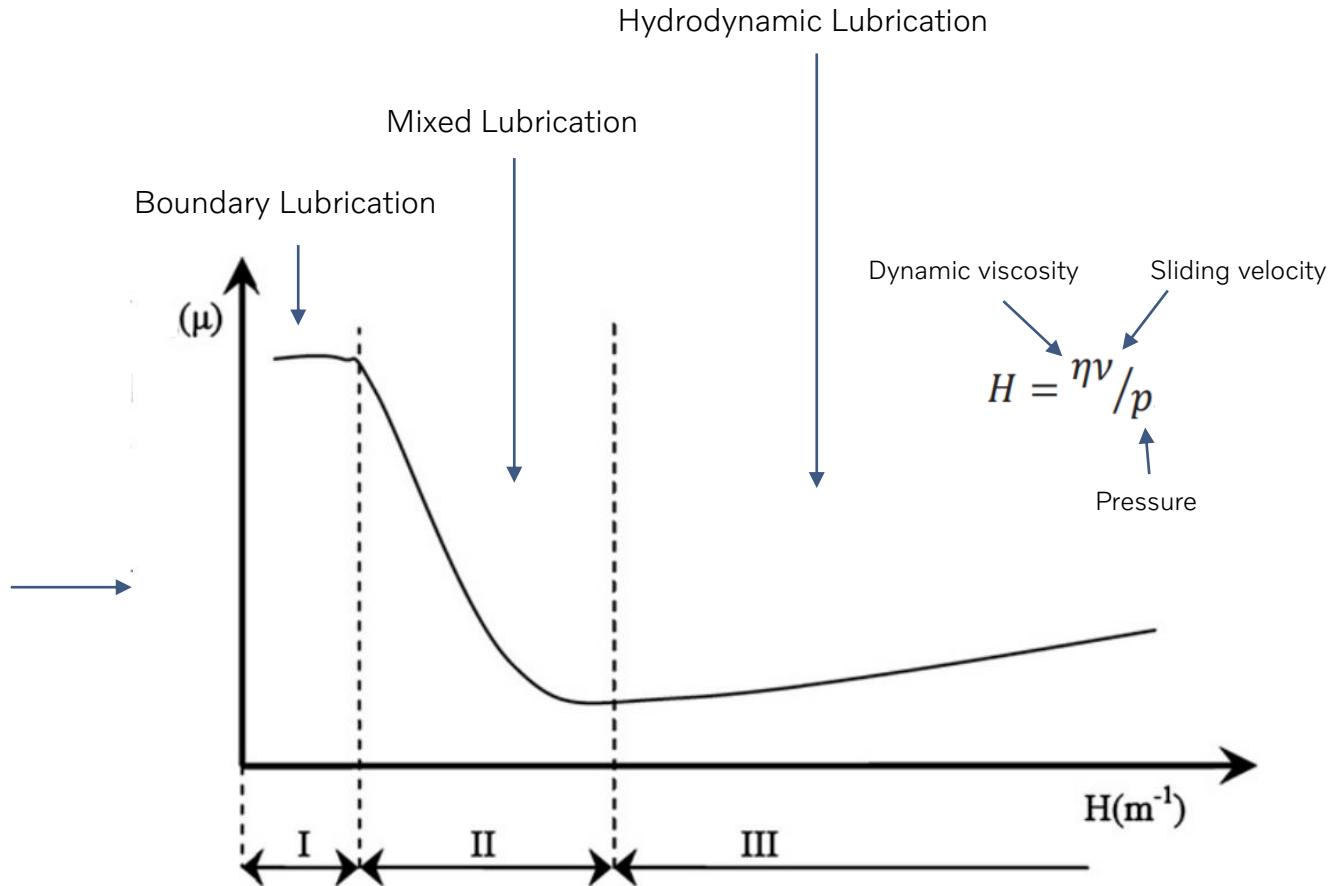
# Interply friction: State of the art

Stacked shell model in contact



State of the art friction models capture dependency of

- Pressure
- Viscosity
- Slip rate

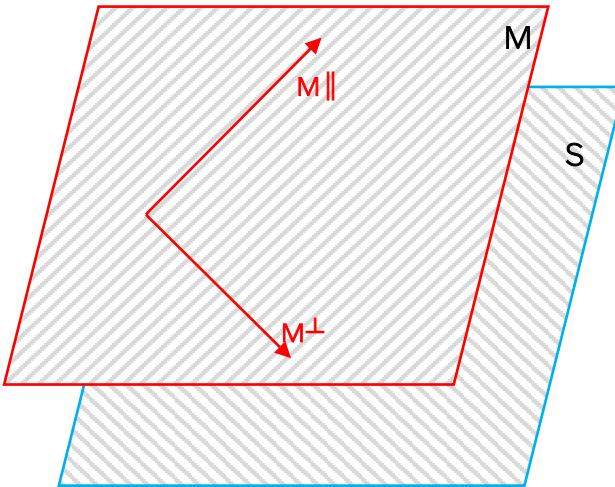


# Interply friction

What about the effect of fiber orientations?

$M_{\parallel}$  is the primary material direction in Surface M

$S_{\parallel}$  is the primary material direction in Surface S



Interply friction has a dependency on fiber angles in consecutive layers!

v o l v o

DYNA  
MORE  
NORDIC

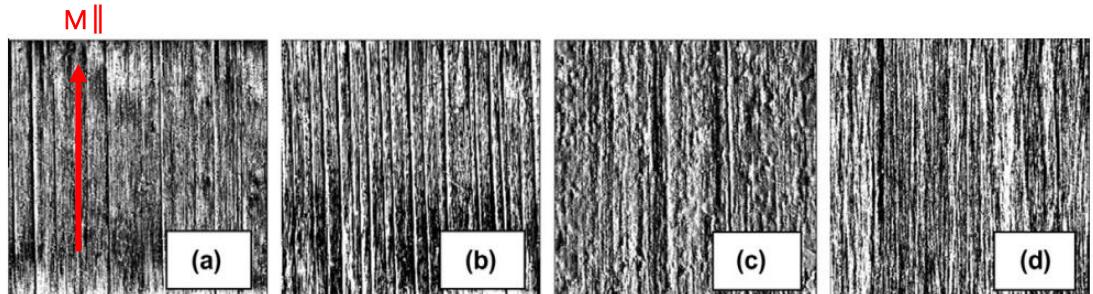
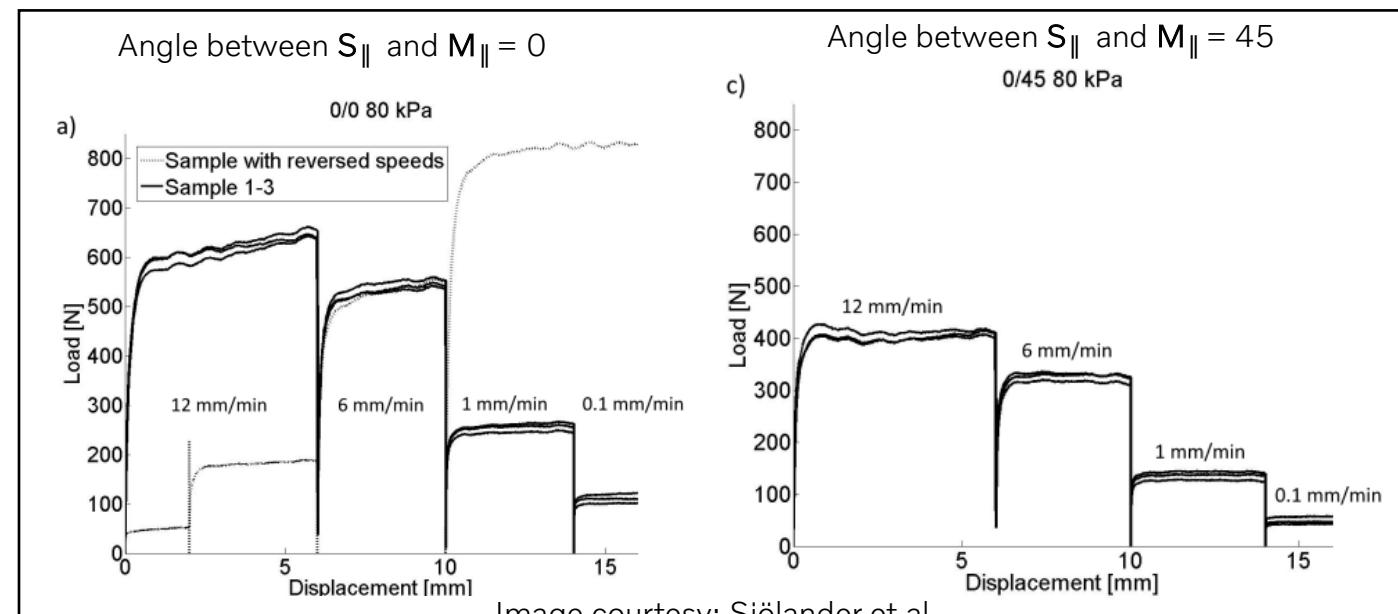


Image courtesy of Larberg et al[5] showing surface topologies of different prepgs at elevated temperatures



# Orthotropic friction

v o l v o

\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_MORTAR\_ORTHO\_FRICTION

Implemented by Thomas Borrval, DYNAmore Nordic

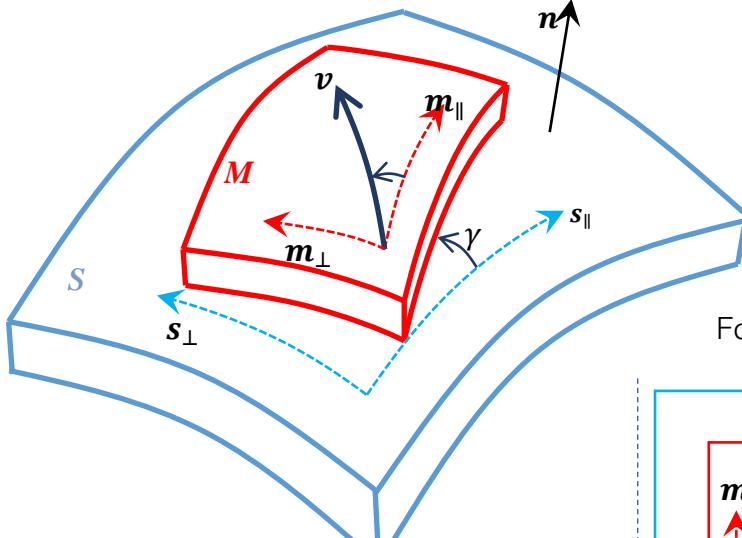


Image from User manual[1]

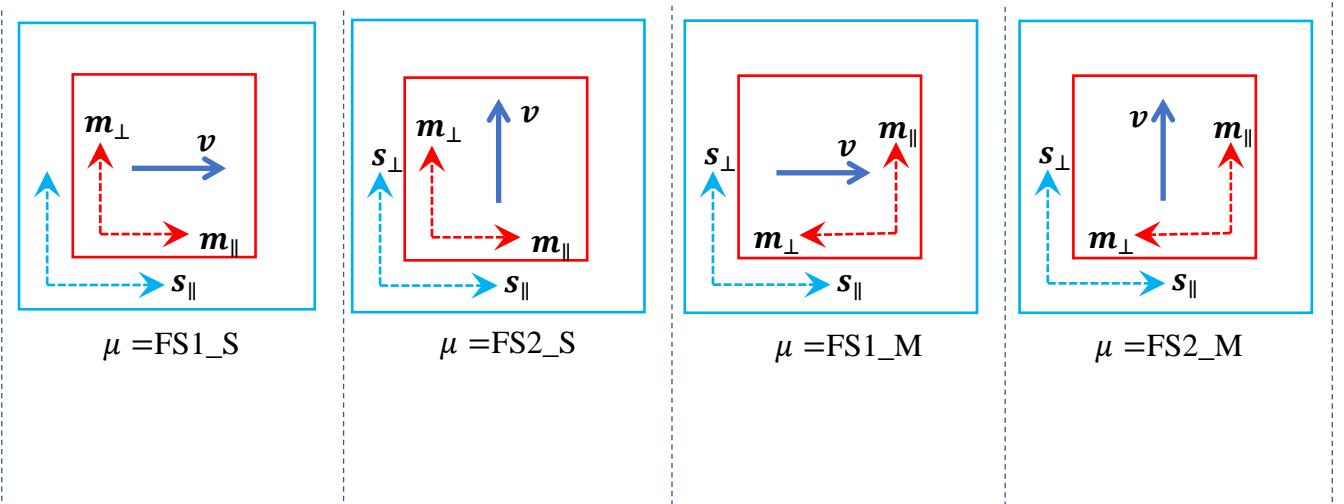
In order to capture the effect of pressure and slip rate, each of these friction values can be represented by

\*DEFINE\_TABLE\_2D\_TITLE

## Parameters that affect Interply friction

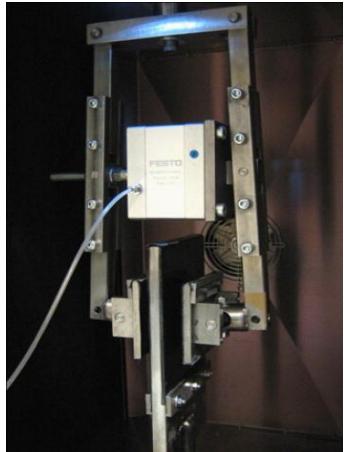
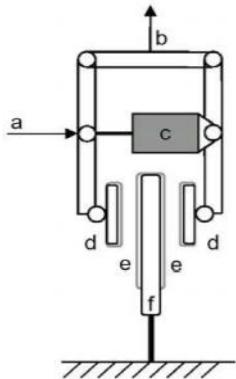
- Difference in primary orientation of the consecutive layers ( $\gamma$ )
- Direction of sliding ( $v$ )
- Slip rate
- Consolidation pressure

For a given pressure and slip rate, the orthotropy in friction is characterized by 4 values

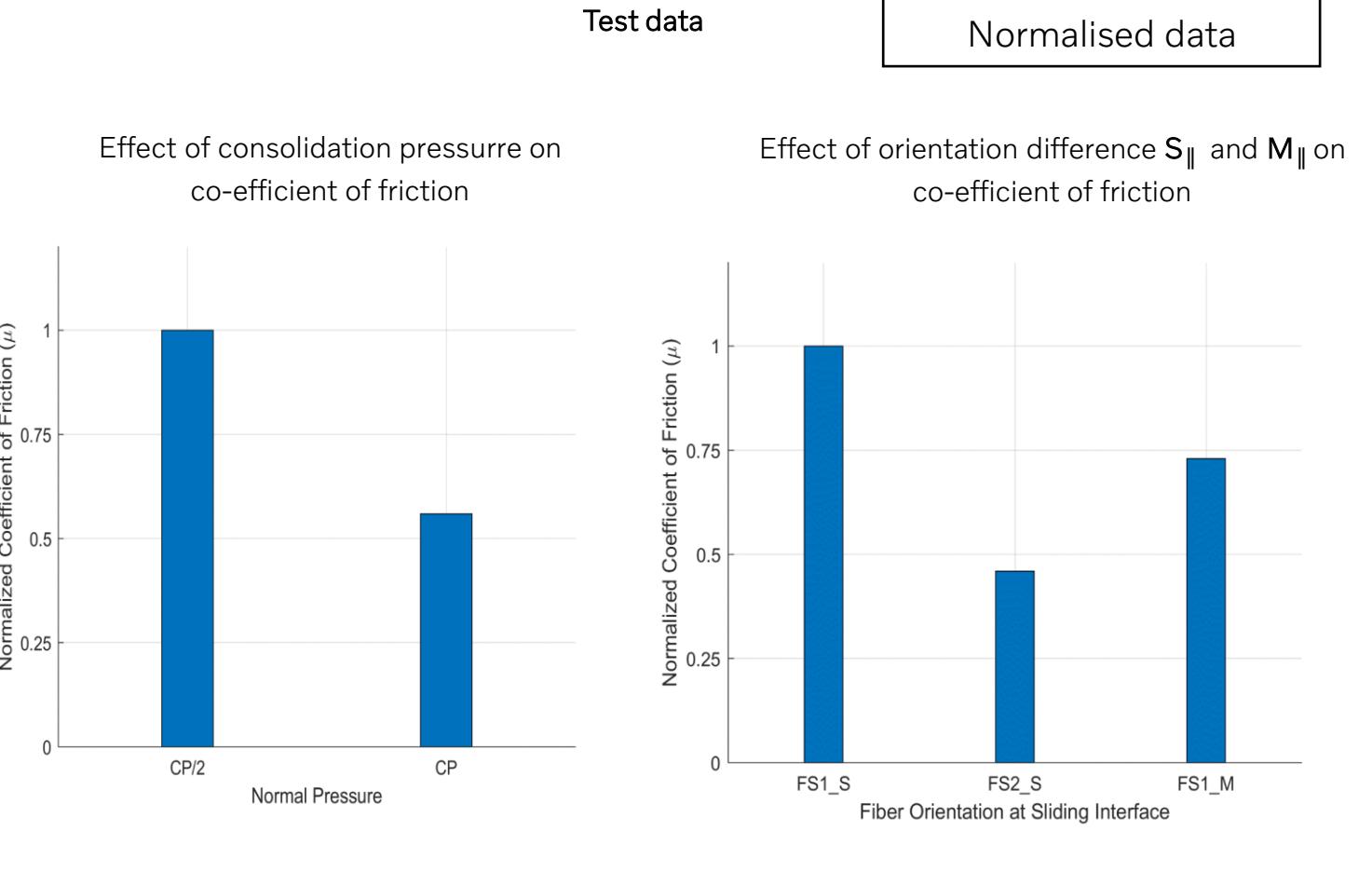


# Material testing

Friction testing rig at KTH



Courtesy of Larberg et al[5]



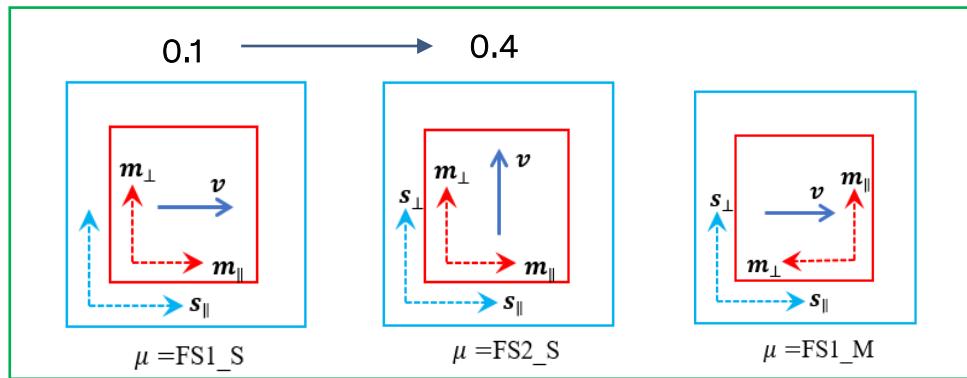
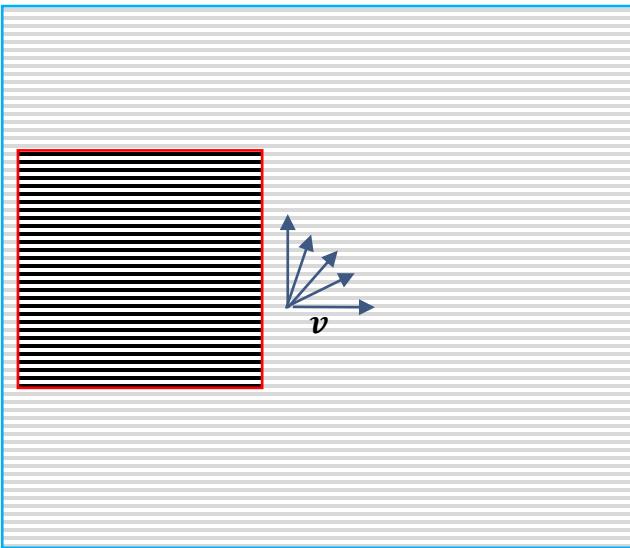
# Example: Interply sliding

v o l v o

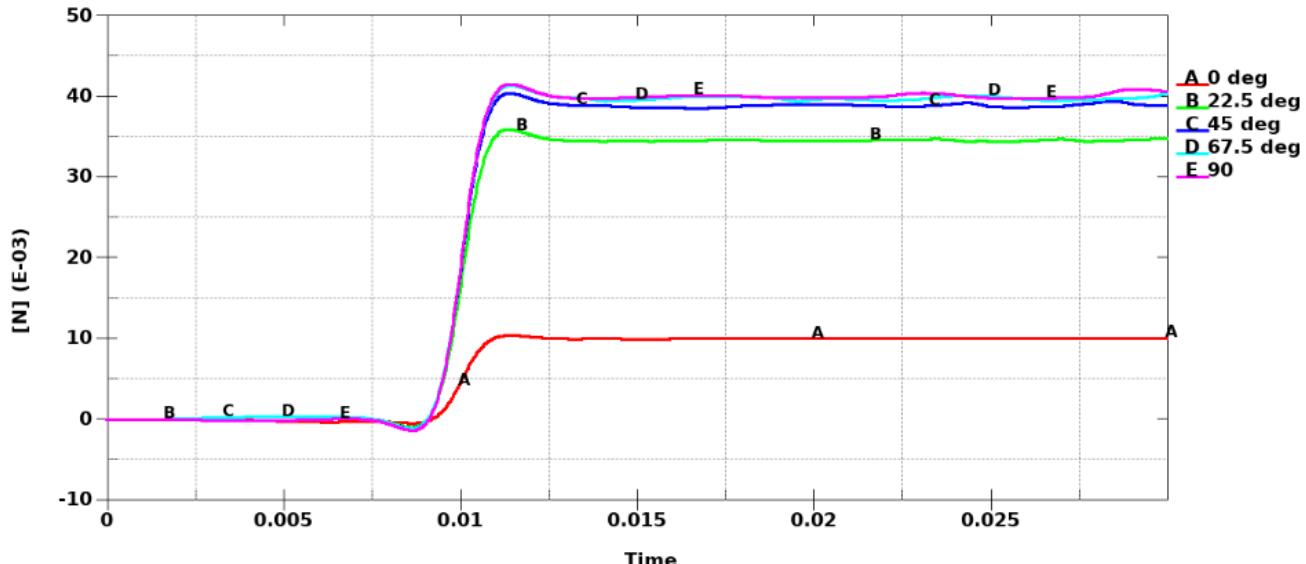
DYNA  
MORE  
NORDIC



Angle between  $S_{\parallel}$  and  $M_{\parallel}$  = 0

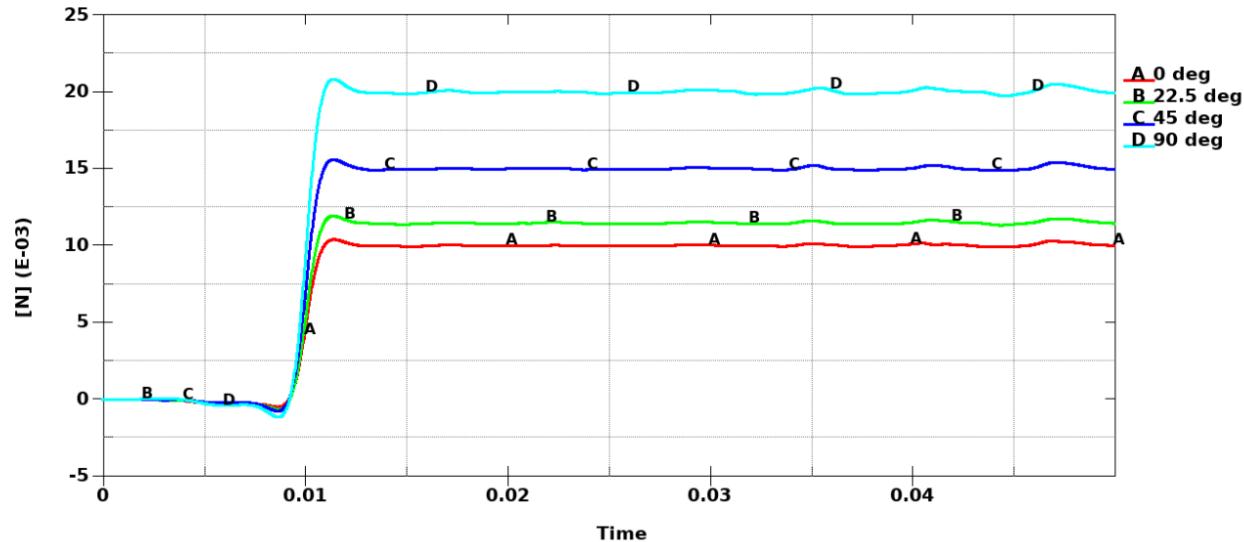
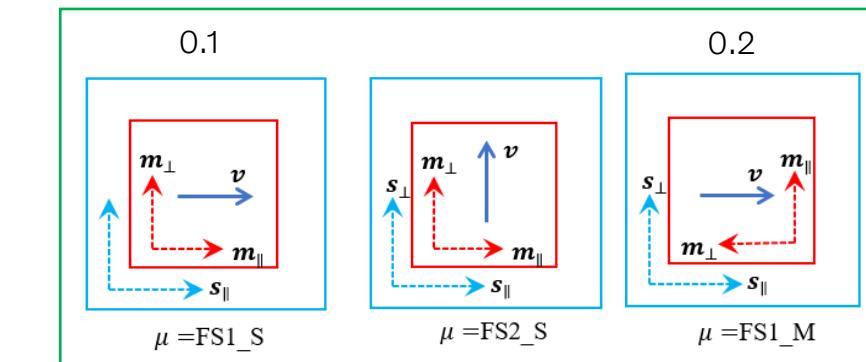
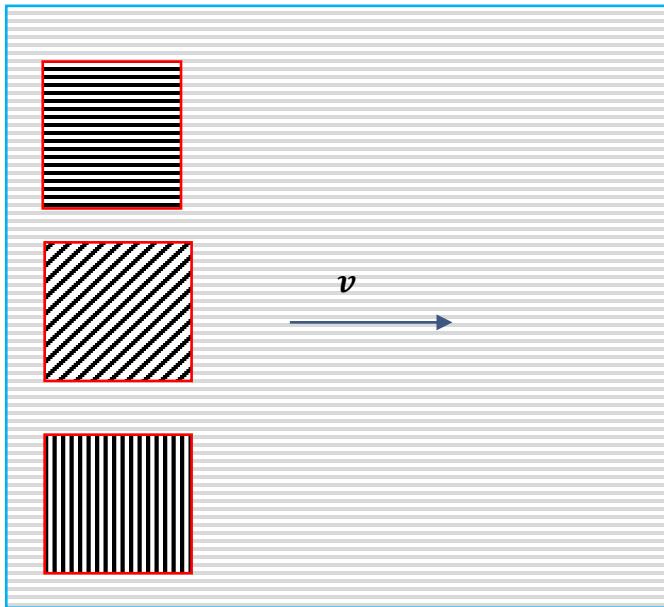


Changing the direction of sliding while keeping the pressure and speed constant.



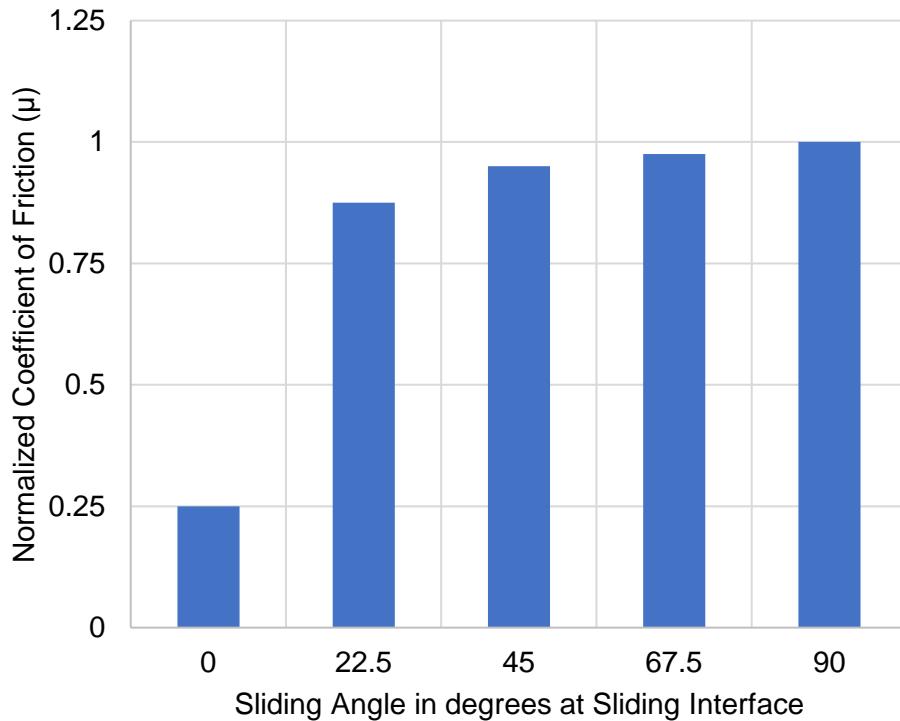
# Example: Interply sliding

Angle between  $S_{\parallel}$  and  $M_{\parallel}$  varied from  $0^0$  to  $90^0$

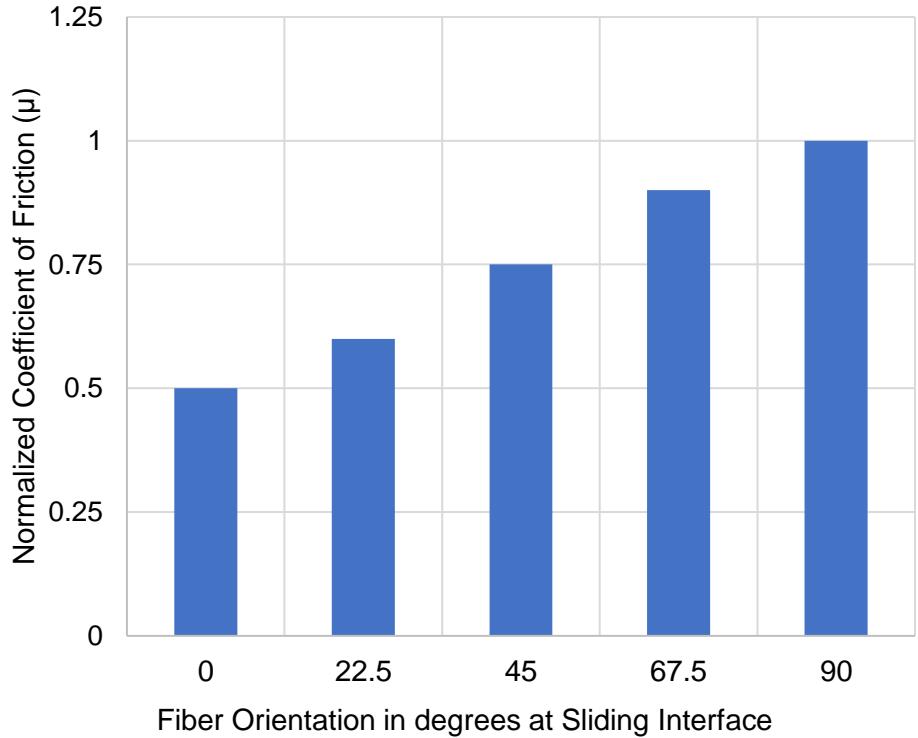


# Interpolation

Elliptical interpolation for fixed angle  
between  $S_{\parallel}$  and  $M_{\parallel}$  with varying  $\nu$



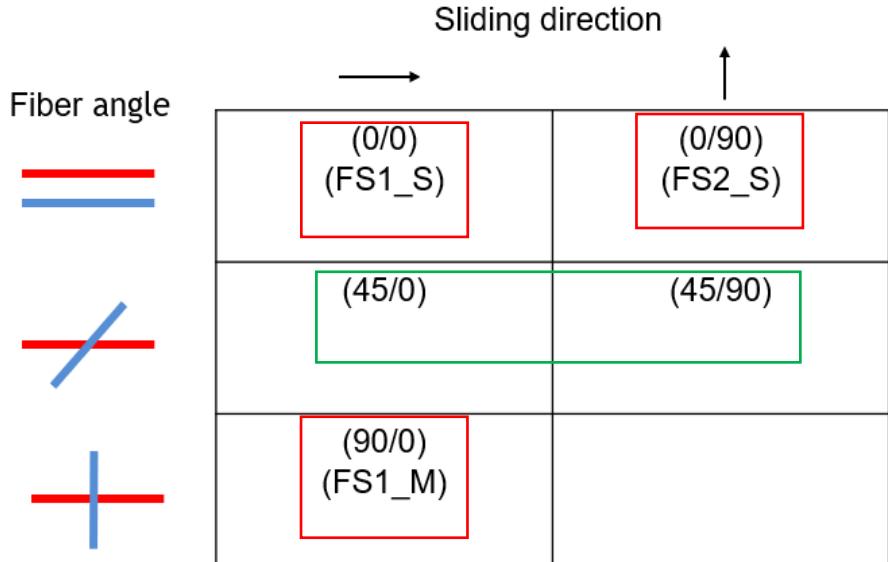
Linear interpolation for varying angle  
between  $S_{\parallel}$  and  $M_{\parallel}$  with fixed  $\nu$



# Correlation to physical tests

Experimentally obtained: Input values

Interpolated values



Normalised data

Sliding interface	Physical test	Simulation
(0/0)	1	1
(0/90)	0.47	0.47
(45/0)	0.72	0.87
(45/90)	0.63	0.61
(90/0)	0.74	0.74

Satisfactory correlation is observed for both defined and interpolated cases.

# **Virtual Demonstrator**

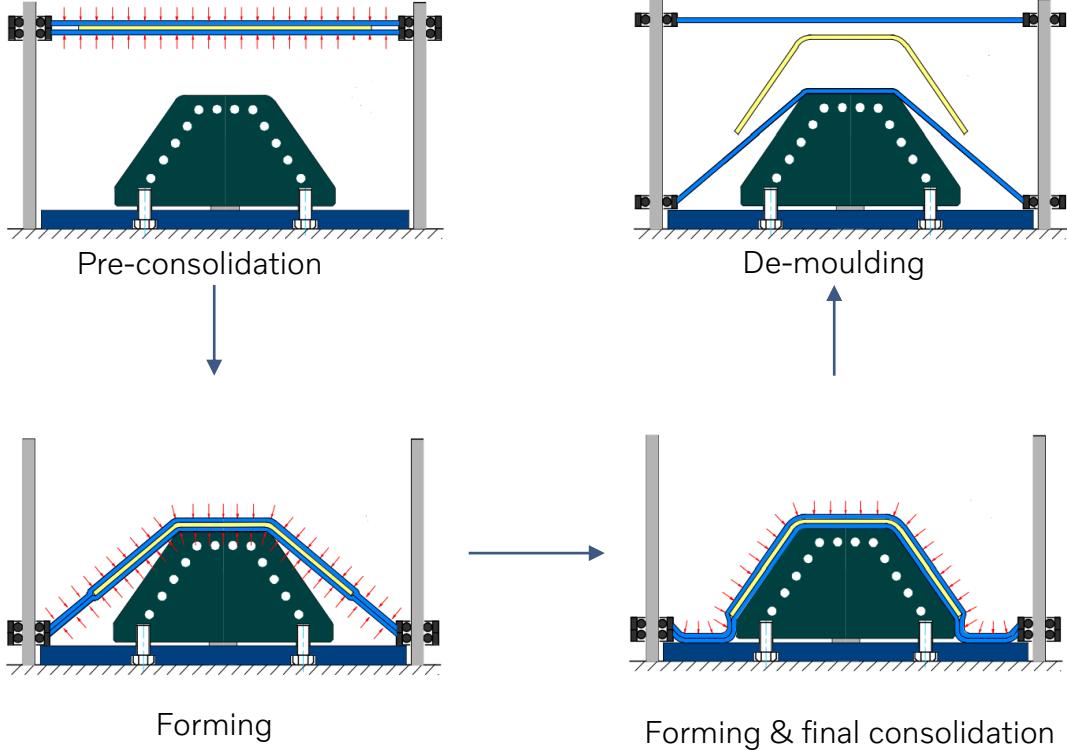


# Virtual demonstrator

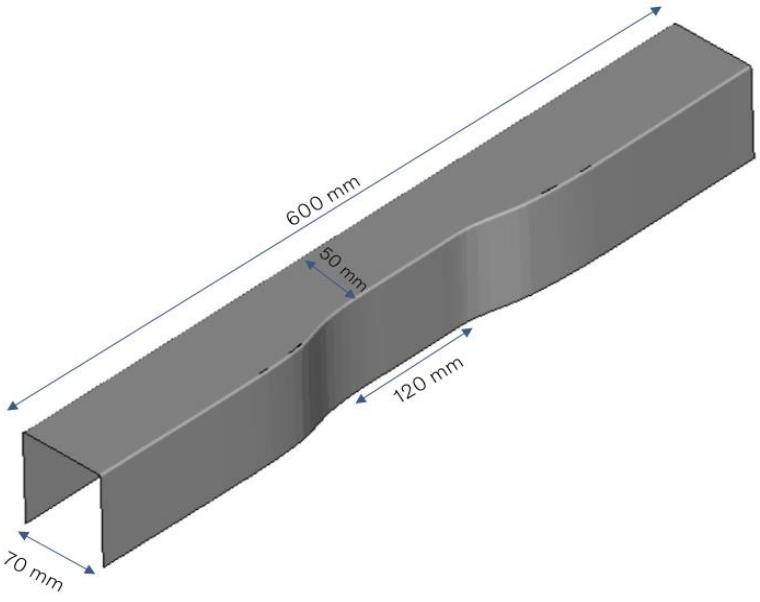
v o l v o



Process: Vacuum assisted Diaphragm Forming



Demonstrator geometry



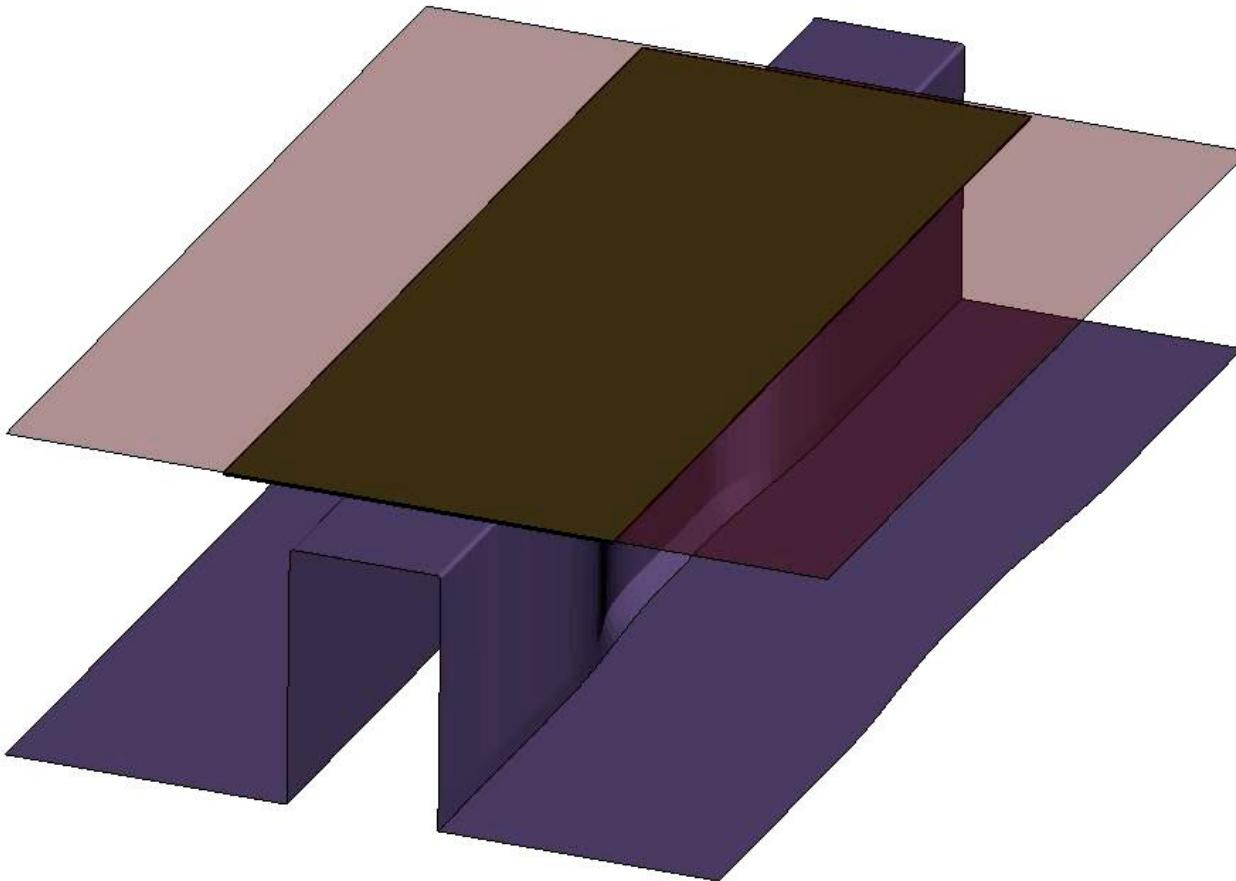
Joggled beam to trigger defects

Stacking sequence → UD plies →  $(+45/-45/0/0)_s$

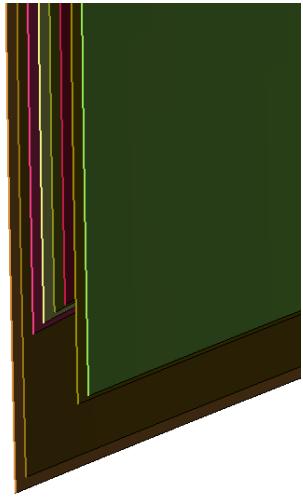
V O L V O

# Diaphragm forming

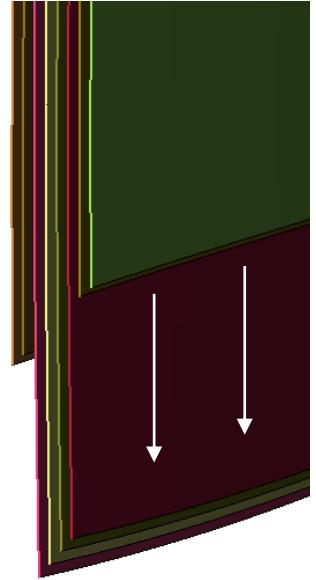
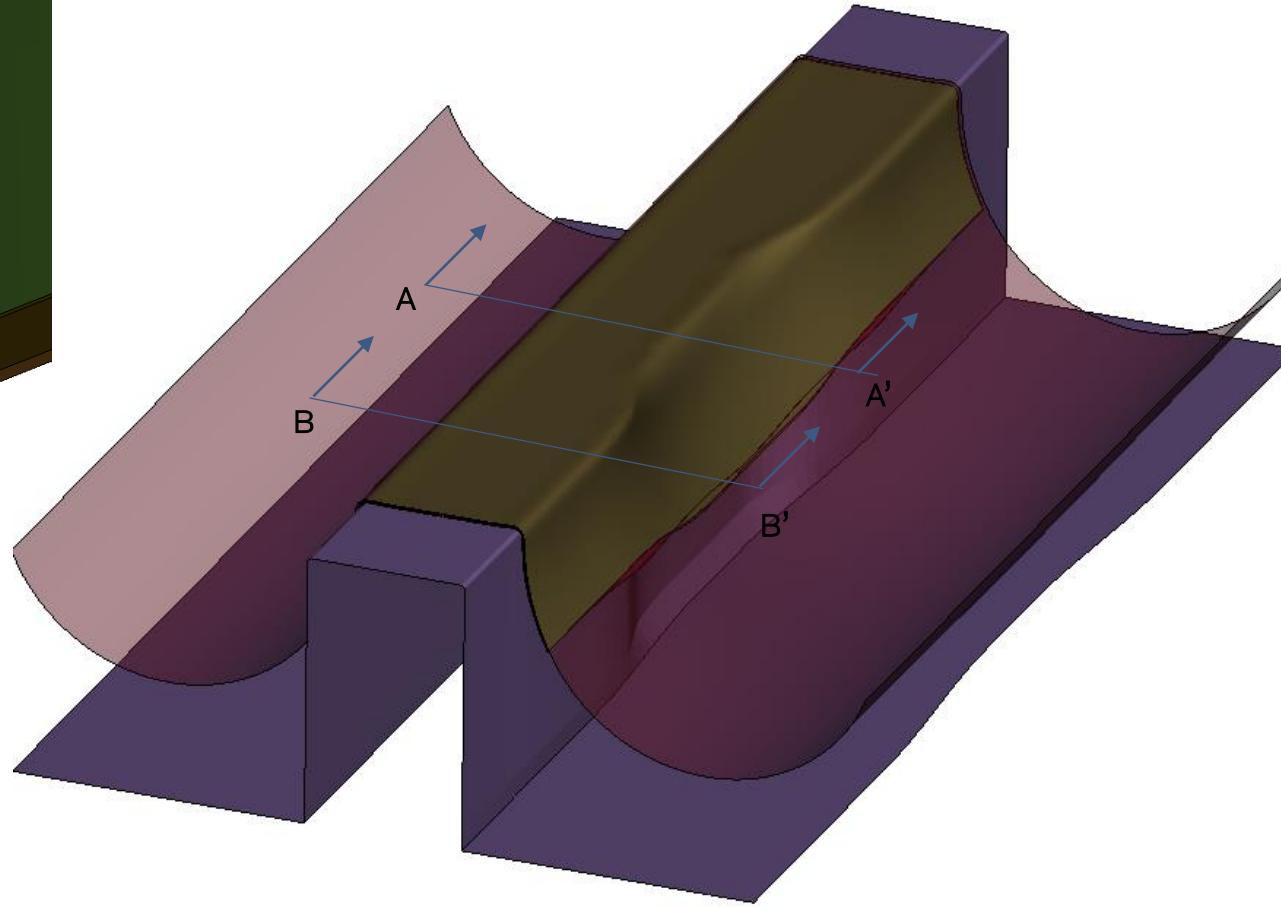
2:d3plot : LS-DYNA Keyword deck by LS-PrePost : STATE 1 ,TIME 0.0000000E+00



# Diaphragm forming

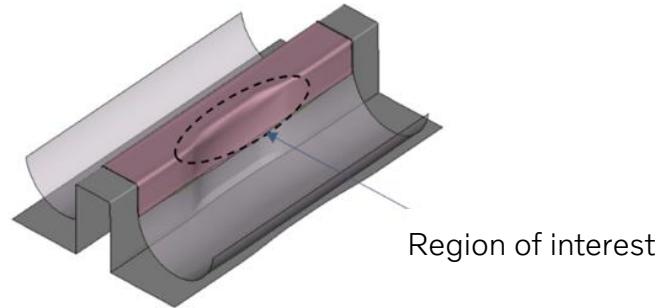


Section A-A'



Section B-B'

# Demonstrator : Contact benchmark



\*CONTACT\_AUTOMATIC  
\_SURFACE\_TO\_SURFACE\_MORTAR



No wrinkles

\*CONTACT\_AUTOMATIC  
\_SURFACE\_TO\_SURFACE\_MORTAR  
with pressure dependency



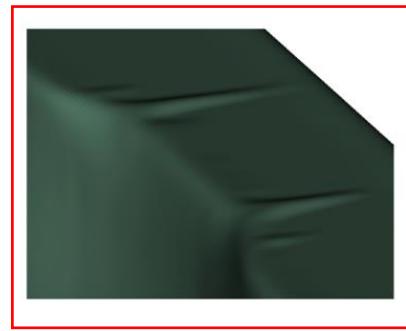
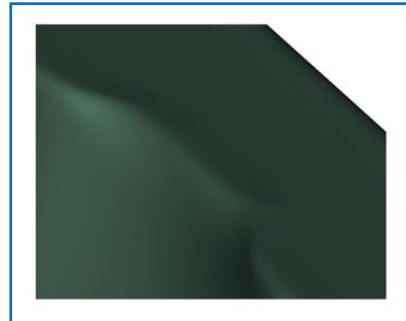
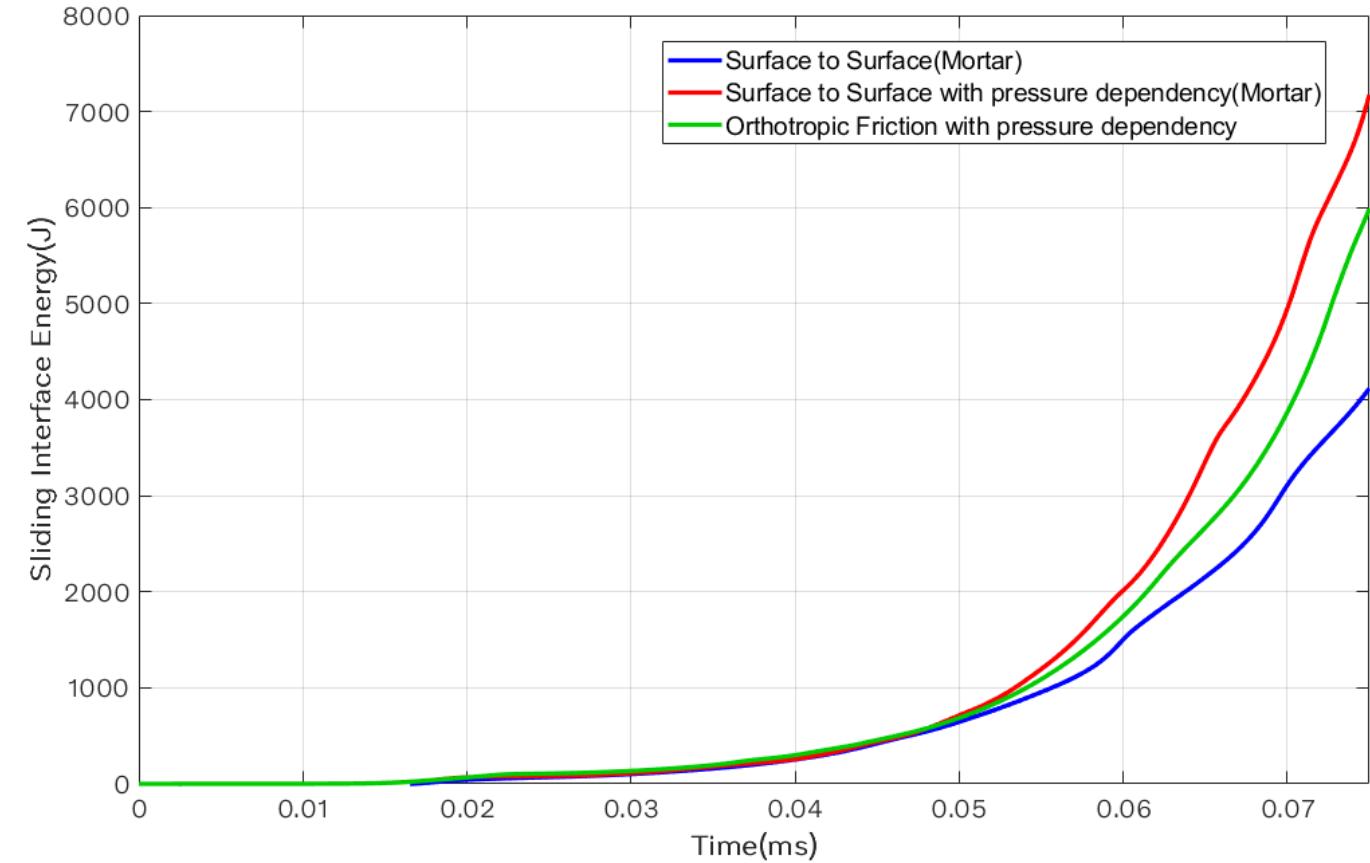
Large wrinkles

\*CONTACT\_AUTOMATIC  
\_SURFACE\_TO\_SURFACE\_MORTAR\_  
ORTHO\_FRICTION



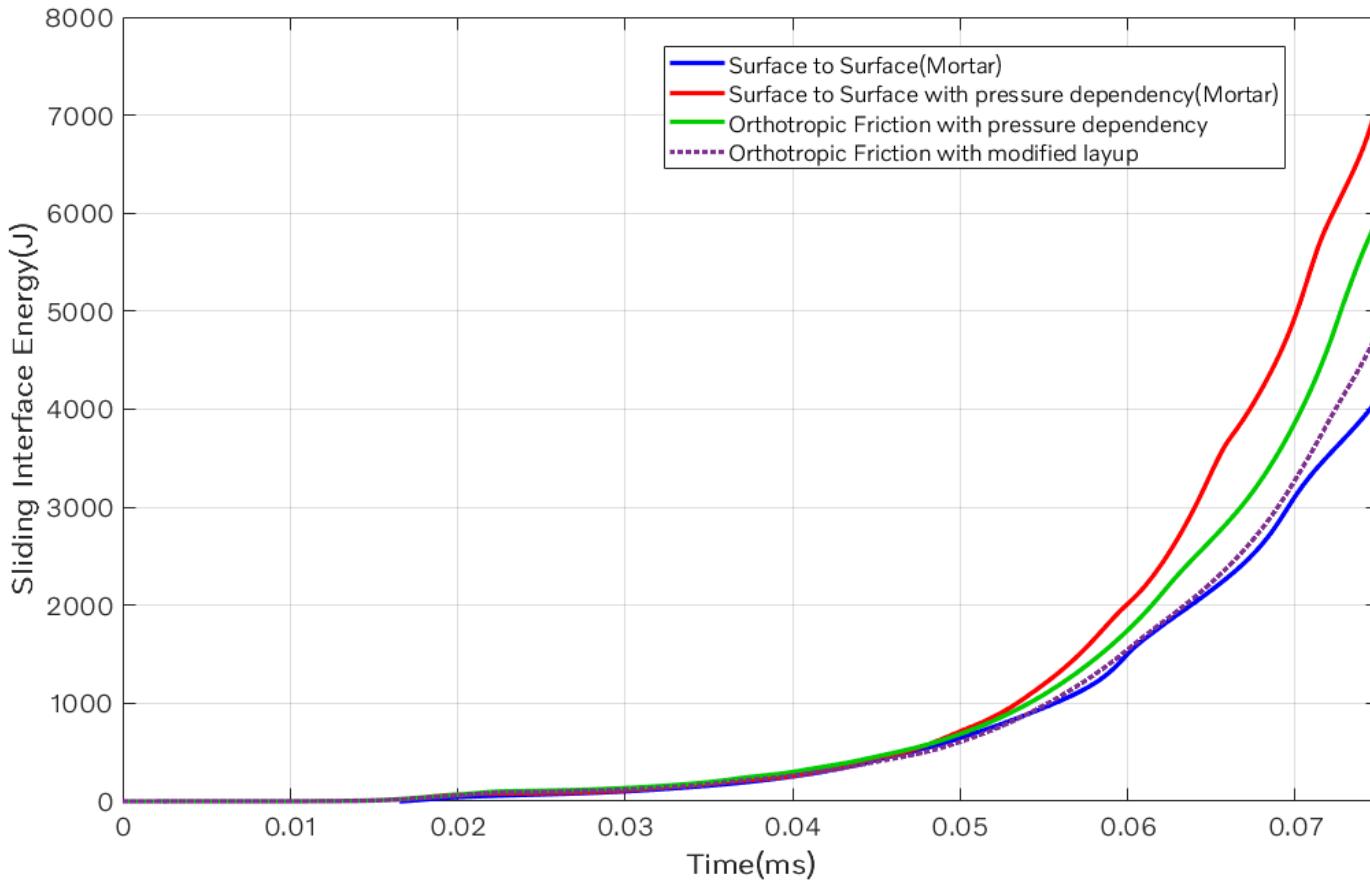
Minor wrinkles

# Sliding interface energies

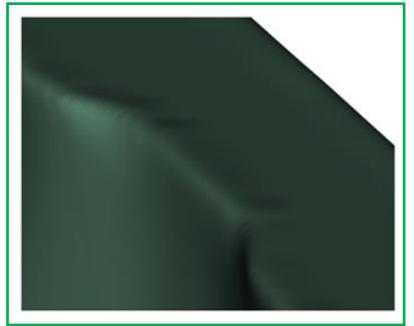


# Sliding interface energies

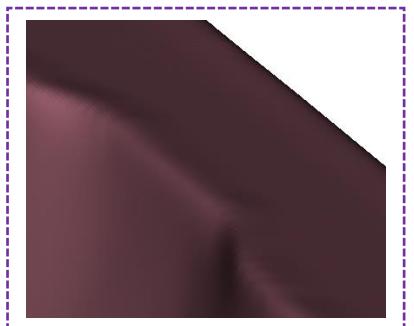
Effect of changing stacking sequence



Lay-up:  $(+45/-45/0/0)_s$



Lay-up:  $(0/90/0/90)_s$



# Summary

- Orthotropic friction with pressure dependence is implemented in a Mortar contact in LS-Dyna
- The model is evaluated against physical tests and shows satisfactory correlation
- Virtual trials show were used to compare different contact formulations.

## Next steps

- Investigate effects of different prepreg architectures
- Correlation to physical tests on a Body-in-white component.

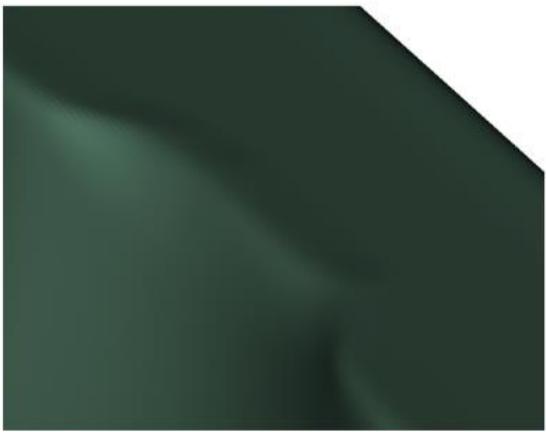


Thank you!

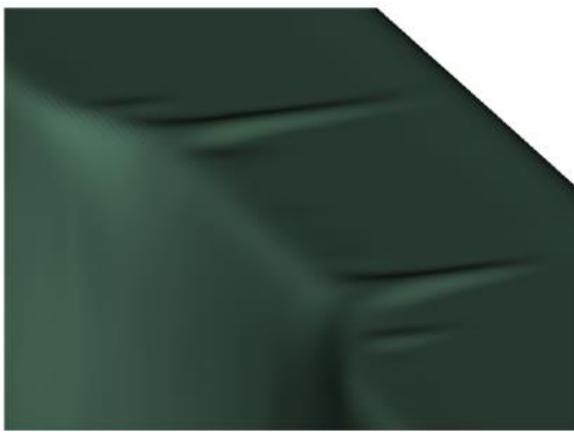


# References

- [1] LS-DYNA® KEYWORD USER'S MANUAL, VOLUME I, 05/06/21 (r:13750), LS-DYNA Dev, LIVERMORE SOFTWARE TECHNOLOGY (LST), AN ANSYS COMPANY.
- [2] Sjölander J: PhD dissertation, "Improving Forming of Aerospace Composite Components through Process Modelling", KTH 2018, ISBN 978-91-7729-881-6.
- [3] Sun J, Li M, Gu Y, Zhang D, Li Y, Zhang Z. Interply friction of carbon fiber/epoxy prepreg stacks under different processing conditions. *Journal of Composite Materials*. 2014;48(5):515-526. doi:10.1177/0021998313476320
- [4] Dutta A.: MSc thesis, "An experimental investigation of interply shear in fast curing composite prepgs.", KTH 2020.
- [5] Larberg Y, Åkermo M.: "On the interply friction of different generations of carbon/epoxy prepreg systems", *Composites Part A: Applied Science and Manufacturing*, Volume 42, Issue 9, 2011, Pages 1067-1074, ISSN 1359-835X,
- [6] Larberg Y. Forming of stacked unidirectional prepreg materials [Internet] [PhD dissertation]. [Stockholm]; 2012. (Trita-AVE). Available from: <http://urn.kb.se/resolve?urn=urn:nbn:se:kth:diva-106269>
- [7] Hallander P. Towards defect free forming of multi-stacked composite aerospace components using tailored interlayer properties [Internet] [PhD dissertation]. [Stockholm]; 2016. (TRITA-AVE). Available from: <http://urn.kb.se/resolve?urn=urn:nbn:se:kth:diva-185694>



A  
No-out of plane wrinkle



B  
Large wrinkle



C  
Minor wrinkle

